



2011 SEAri Annual Research Summit

Research Report from the Field

"MATE Applied in Industry: A Satellite Constellation Trade Study"

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Cambridge, MA

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Outline

- Introduction
 - MATE in an industry satellite application
- Method:
 - Populating the tradespace by iteration using expert tools
- Exploring the Tradespace by Attribute
- Multi Attribute Tradespace Exploration (MATE)
- Conclusions





The Mission

- Earth observation constellation with global coverage and requirements for:
 - high observation times
 - frequent revisits
 - fast downlinks
 - with high reliability for multiple points on Earth
- How these attributes should be traded off versus each other or versus costs?





The Team

- How do we engage knowledge from groups of domain experts:
 - Orbital analysts
 - Mission analysts
 - Spacecraft bus engineers
 - Payload specialists
 - Launch analysts & contractors
 - Ground station engineers
 - Operations engineers





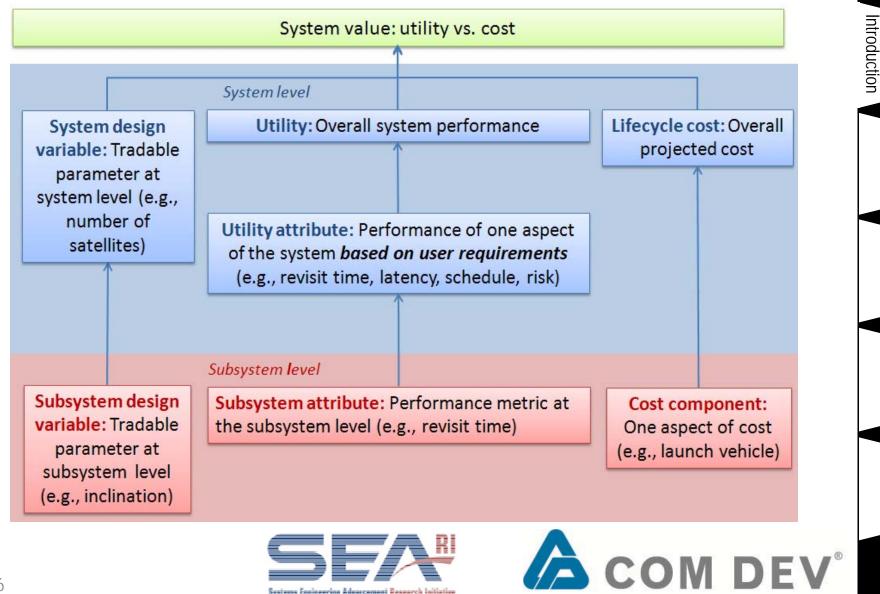
Application of MATE

- Space missions have competing design alternatives that span tradeoffs across multiple subsystems
 - It is not straightforward to select from design alternatives
- Customer requested tradeoff studies
 - However, previous tradeoff studies had been individual performance tradeoffs within a single subsystem – no comprehensive system level trades
- No system level performance model available
- How do we consider interactions across multiple subsystems while mobilizing domain expert knowledge?



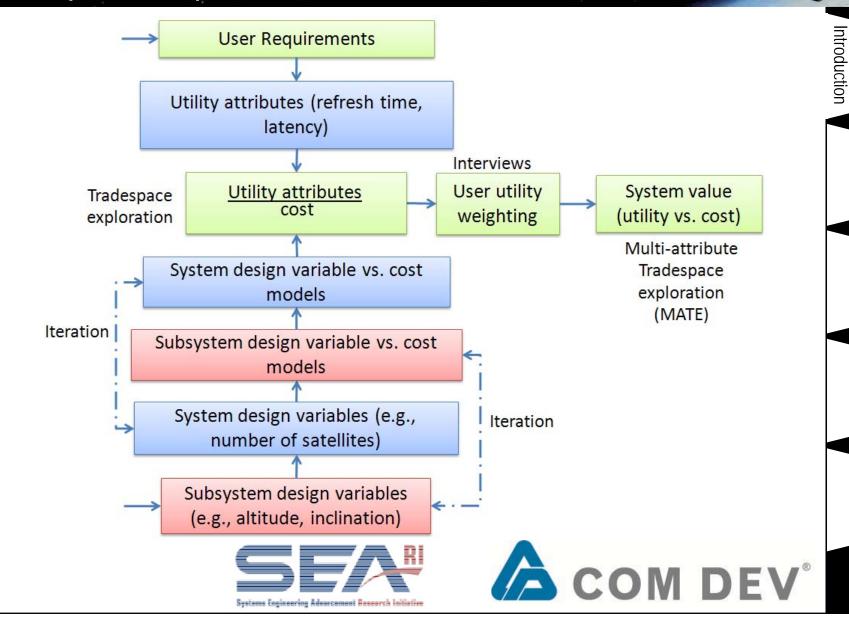


Terminology & Hierarchy



Systems Engineering Advarcement Research Initial

Tradespace Exploration Flow

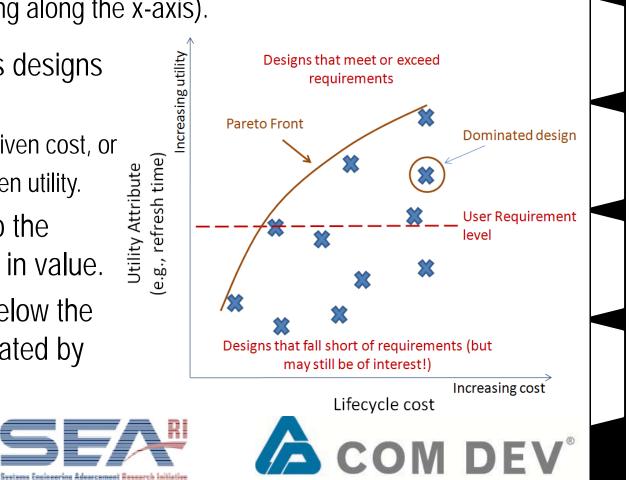


Example Tradespace

- Increasing *value* is composed of:
 - increased utility (increasing along the y-axis, and

Jtility Attribute

- lower cost (decreasing along the x-axis).
- Pareto Front represents designs that have:
 - the highest utility for a given cost, or
 - the lowest cost for a given utility.
- Designs falling closer to the Pareto Front are higher in value.
- Designs falling father below the Pareto Front are dominated by higher value designs.



Method

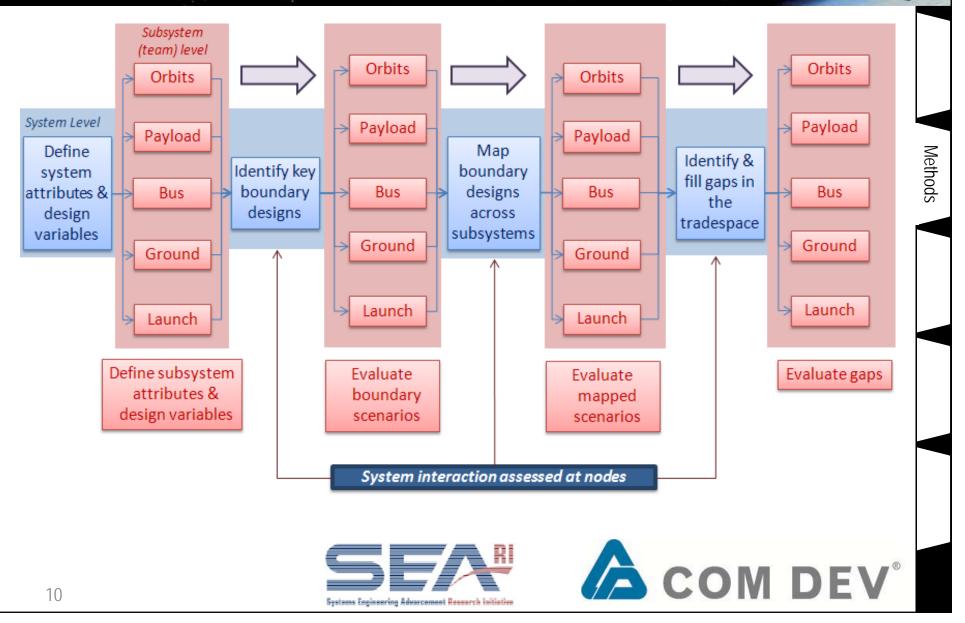
- Subsystem level optimizations performed in parallel by domain experts using their own tools.
- Allows independent partitioning of the design space by expertise (e.g., orbit, ground, launch, payload, etc.).
- Interactions between subsystems assessed and propagated at system level using key nodal checkpoints.
- Assembled system level tradespace automatically captures individual subsystem level trades (i.e., it is pre-filtered).





Methods

Populating the Tradespace by Iteration



System Level Interactions

- 1. Evaluate interactions at system level: i.e., how do subsystem design variables influence system level utilities and costs?
- Map subsystem-generated design variables across subsystems (possible full factorial expansion, e.g., 5 orbit x 3 ground station x 2 launch = 5 x 3 x 2 = 30, but overlaps are likely).
- 3. Potential for pruning the initial Tradespace.
- 4. Add missing "gap" designs based on exploration of initial Tradespace – especially when two or more subsystems share design variable elements (e.g., launch analysis suggests cost savings from implementing an orbital configuration).





Methods

Expert Group Communication

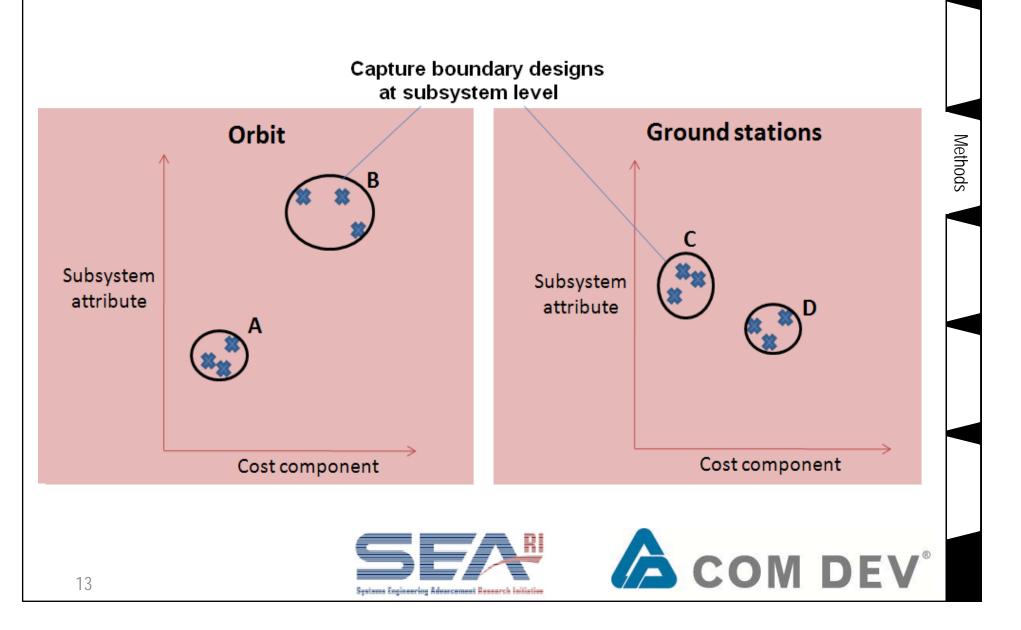
- Minimal communication between expert groups is required to ensure that designs are feasible (though not necessarily efficient) across subsystems.
- No need to account for interactions at the subsystem level.
- Note that two subsystems may not produce the same result even with the same goal.



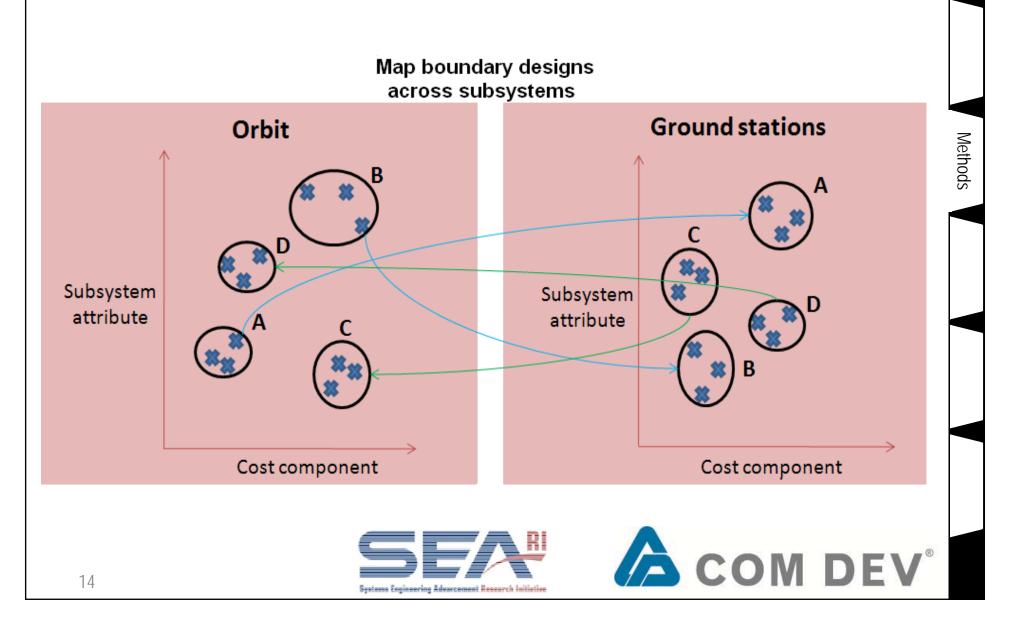


Methods

Capture Boundary Designs

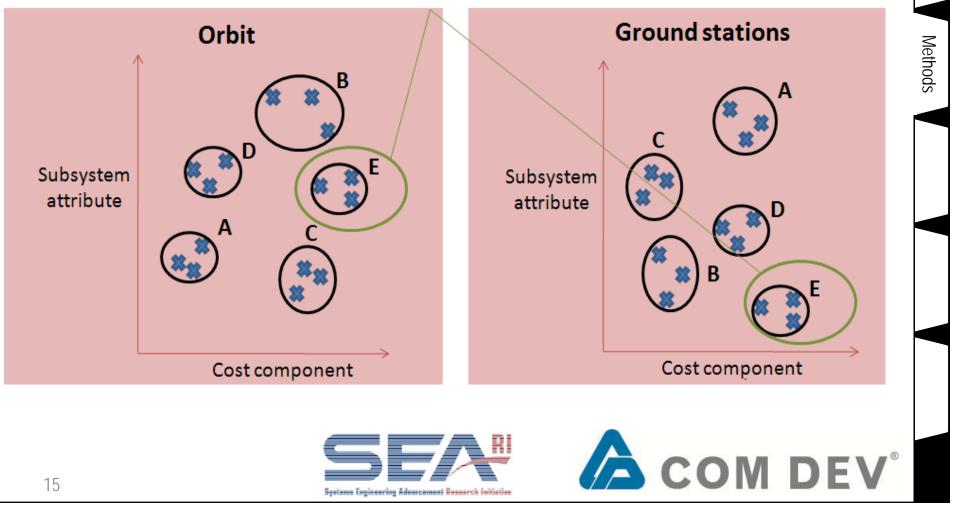


Map Across Subsystems

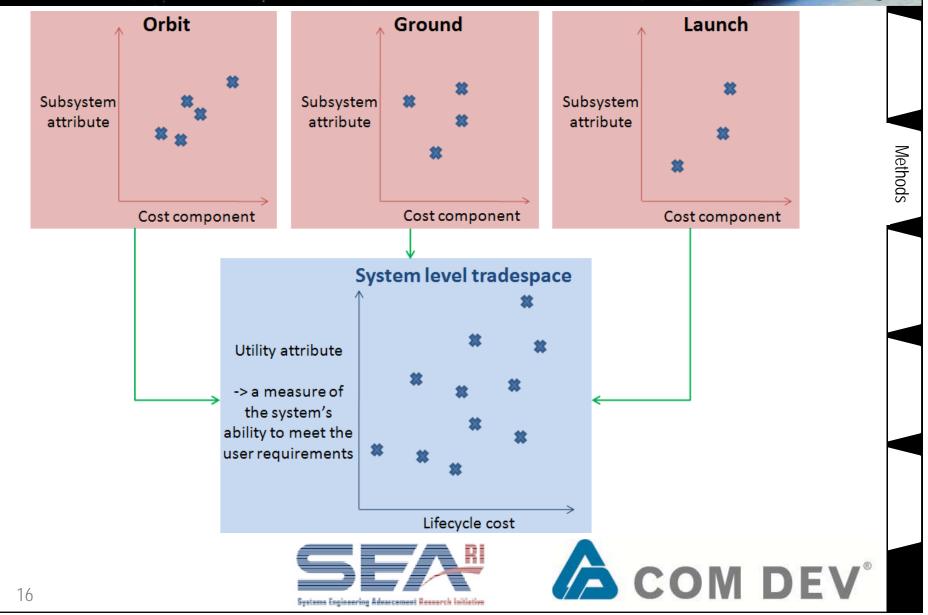


Fill Gaps in Tradespace

Fill gaps in tradespace: Design E not captured as a boundary design



Build System Level Tradespace



Satellite Constellation Attributes

Utility attribute	Contributing subsystems	
	• Orbit	
Refresh time globally	• Launch (only as an orbit driver)	
	• Ground	
Patrash time over particular area of	• Orbit	
Refresh time over particular area of	• Launch (only as an orbit driver)	
interest	• Ground	Q
	• Orbit	Attributes & design variables
Time on target	• Launch (only as an orbit driver)	ibute 1 var
	Payload (footprint, duty cycle)	iable
	• Orbit	S
Latanay	• Launch (only as an orbit driver)	
Latency	Payload (data volume and format, downlink rate)	
	• Ground (time to downlink, distribute, & process data)	
	Bus (reliability/availability)	
Redundancy	• Orbit (refresh time with loss of spacecraft)	
	• Ground (refresh time with loss of ground station)	





Satellite Constellation Design Variables

System design variables	Subsystem design variables	Associated cost drivers
Spacecraft/bus	 Number of spacecraft Nanosat/microsat/smallsat/hosted payload Controlled/uncontrolled 	Design & buildLaunchOperations
Orbital characteristics	 Orbit planes Eccentricities Separation Special cases (e.g., sun-synchronous) 	LaunchPropulsion & delta-v
Launch vehicle	 Number of satellites per launch Primary vs. secondary payload Replacement availability 	Launch vehiclesLaunch operations
Payload	• Selection of payload for mission	 Equipment design & build cost Operations
Downlink	FrequencyBandwidthGeographic availability	Equipment design & build costRegulatory issues
Redundancy, reliability, & replacement strategy	 On-orbit (hot or cold) spares vs. replacement Spacecraft reliability (expected lifespan) 	• Initial cost vs. replacement cost
Ground station	 Number and placement of stations Build vs. buy Stationary vs. mobile 	 Equipment design & build cost Start-up vs. operational costs
Data processing & handling	StorageSecurity	Start-up costsOperational costs





Attributes & design variables

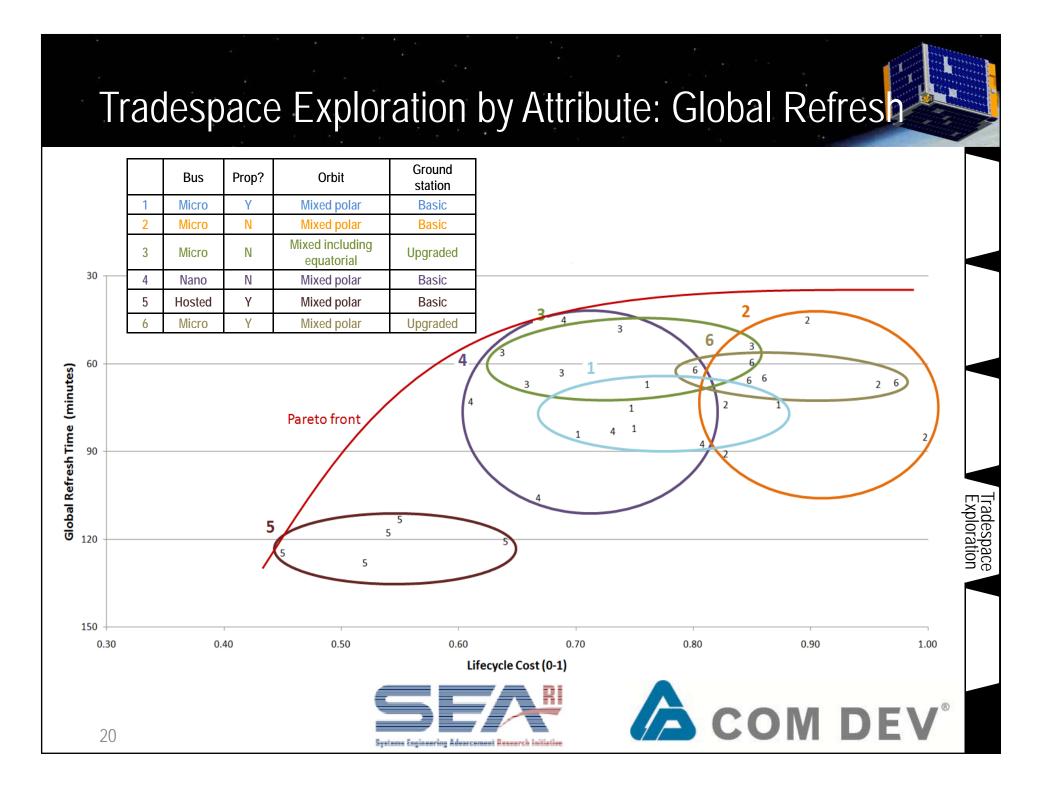
Example Designs

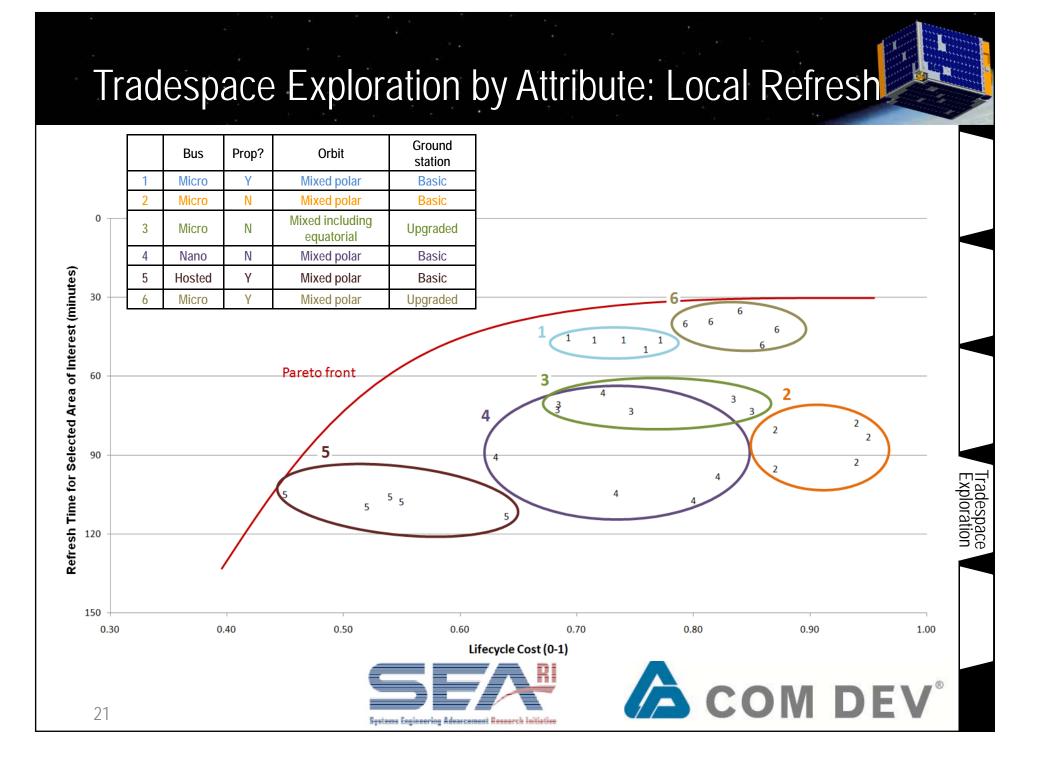
Design number	Bus	Propulsion?	Orbit type	Ground station configuration
1	Microsatellites	Y	Mixed polar	Basic
2	Microsatellites	Ν	Mixed polar	Basic
3	Microsatellites	Ν	Mixed including equatorial	Upgraded
4	Nanosatellites	Ν	Mixed polar	Basic
5	Hosted payload	Y	Mixed polar in historical locations	Basic
6	Microsatellites	Y	Mixed polar	Upgraded

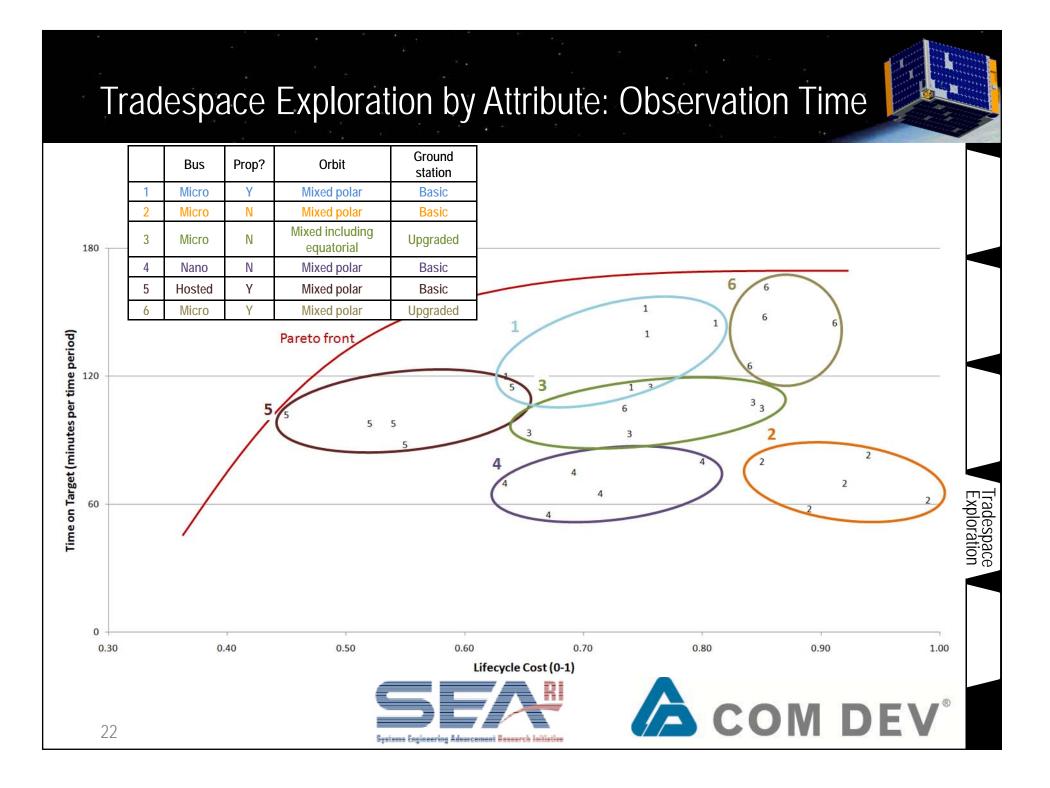


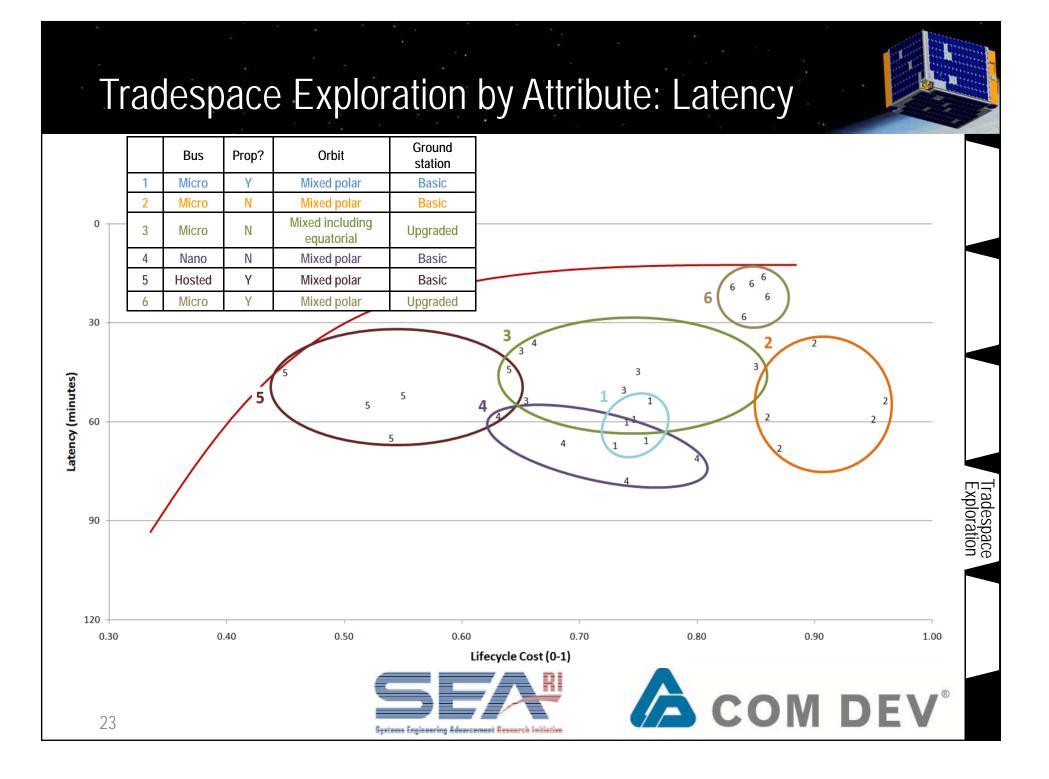


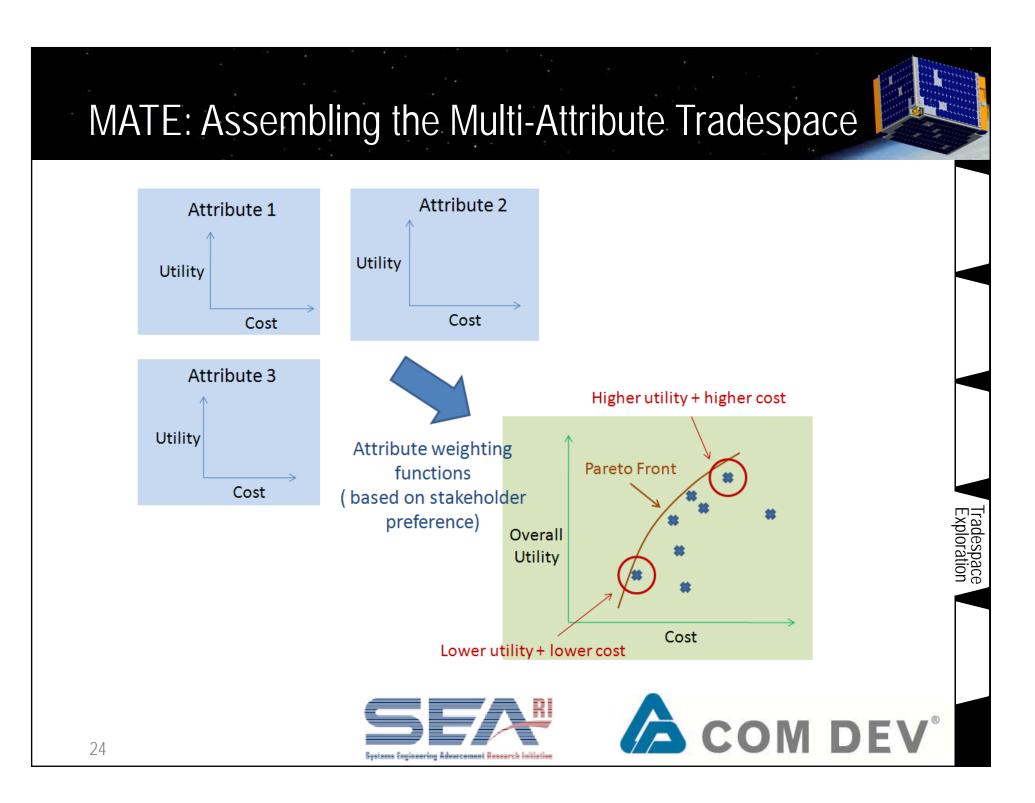
Attributes & design variables





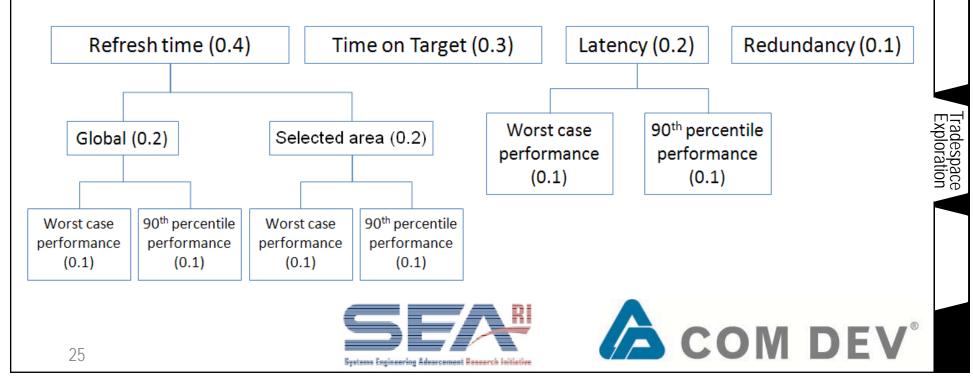


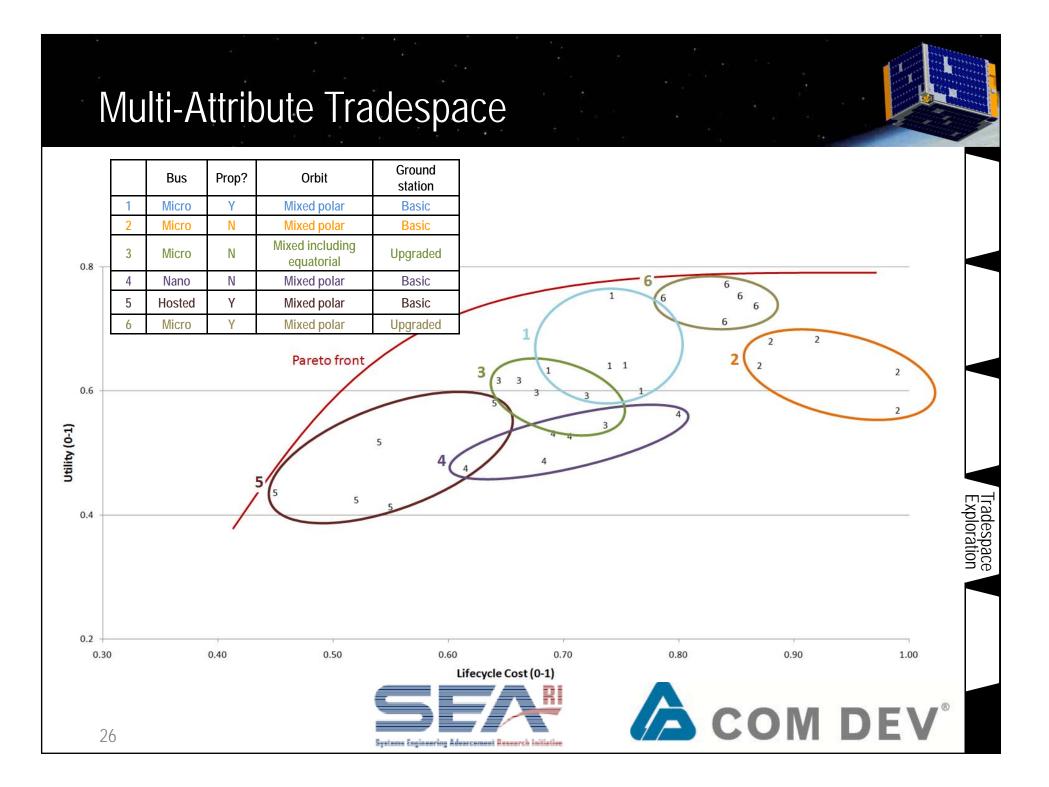




MATE: Utility Weightings

- Conjoint Analysis used to elicit utility weightings:
 - User selects preference from 2 "equal" alternatives of varying attributes multiple times.
 - Difference between nominal attribute rank and true attribute rank allows inference of attribute weightings.





Conclusions: Methodology

- Engaging subsystem domain experts early:
 - Capitalizes on expert knowledge
 - Takes advantage of existing tools
 - Saves time
 - Reduces risk
- Result is a filtered subset containing only high value solutions
 - Performance optimizations at the subsystem level have already been performed in the assembled system level tradespace
- Potential drawback: possible missed solutions in filtered tradespace





Conclusions

Conclusions: Industry Application of MATE

- Allowing domain experts to user their own tools meant that no restructuring was required
- Subsystem groups were not required to be familiar with MATE
- Tasks were discretized and structured hierarchically (with information flow in both directions)
- Customer and company recognized power of methodology (and were more ready to accept it)





Conclusions

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